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AMENDMENTS TO THE CLAIMS

This listing of the claims replaces all prior versions, and listings, of claims in the application:

LISTING OF CLAIMS

- Claim 1. (Original) A casing for a turbofan engine, the engine including at least a fan assembly, a compressor assembly, a combustor assembly and a turbine assembly, the casing comprising:
 - a fan case portion;
 - an intermediate case portion; and
 - a gas generator case portion,
 - wherein the fan case portion, the intermediate case portion and the gas generator case portion are integrally joined together, thereby forming an integral casing.
- Claim 2.(Original) The casing as claimed in claim 1 wherein the fan case portion, the intermediate case portion and the gas generator case portion are made of the same material.
- Claim 3. (Original) The casing as claimed in claim 1 wherein the intermediate case portion further comprises an integral compressor shroud portion and an integral bearing mount portion.
- Claim 4. (Original) The casing as claimed in claim 3 wherein the bearing mount portion is configured to provide integral damping to a shaft bearing.

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- Claim 5. (Original) The casing as claimed in claim 1 wherein the individual fan case portion, the intermediate case portion and the gas generator case portion are fabricated individually and welded together.
- Claim 6. (Original) The casing as claimed in claim 1 wherein the fan case portion, the intermediate case portion and the gas generator case portion are joined together by flangeless connections.
- Claim 7. (Original) A bypass turbofan engine comprising:
 - at least a fan, a compressor, and a gas generator disposed in flow series within the engine, and a bypass airflow defined around at least the compressor and gas generator; and
 - a one-piece casing substantially encasing the fan, compressor and gas generator.
- Claim 8.(Original A turbofan engine as claimed in claim 7 wherein the casing further comprises an integral compressor shroud encircling blade tips of the compressor.
- Claim 9 (Original) A turbofan engine as claimed in claim 8 wherein the casing further comprises an integral bearing seat for directly mounting a compressor shaft bearing to the casing.

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- Claim 10. (Original) A turbofan engine as claimed in claim 9 wherein bearing seat is configured to provide integral damping to the compressor shaft bearing.
- Claim 11. (Original) The turbofan engine as claimed in claim 7 wherein the casing at least partially defines a by-pass air flow passage within the engine.
- Claim 12. (Original) A turbofan engine for an aircraft comprising:
 - a rotating assembly including a propulsive fan portion, a compressor portion, and a gas generator portion, the rotating assembly having an axial length; and
 - a generally tubular casing assembly enveloping the rotating assembly substantially along the axial length thereof and thereby defining a main flow path through the engine, wherein the casing assembly is an integrated single piece.
- Claim 13. (Original) The turbofan engine for aircraft as claimed in claim 12 wherein the casing assembly further comprises a integral shroud section encircling a plurality of compressor blade tips of the compressor portion.
- Claim 14. (Original) The turbofan engine for aircraft as claimed in claim 12 wherein the casing assembly further comprises a integral bearing seat for

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operatively supporting a compressor shaft of the compressor portion.

Claim 15. (Original) The turbofan engine for aircraft as claimed in claim 12 wherein the casing defines at least a portion of a by-pass air duct of the engine.

Claims 16-23 (Cancelled)